GENERIC STALL





- Introduction
- Stall phenomenon
- AoA control
- Stall vs. Approach to stall
- Stall recovery
- Way forward: New procedure









Introduction

- Accidents following failure to recover from stall still occur
- Stalls are frequently performed in flight test by all aircraft manufacturer for development and certification:
 - Unique experience to be shared with operational community
- Objective of this presentation: to make the operational community aware of the need to change the current methods of STALL or APPROACH TO STALL recovery techniques



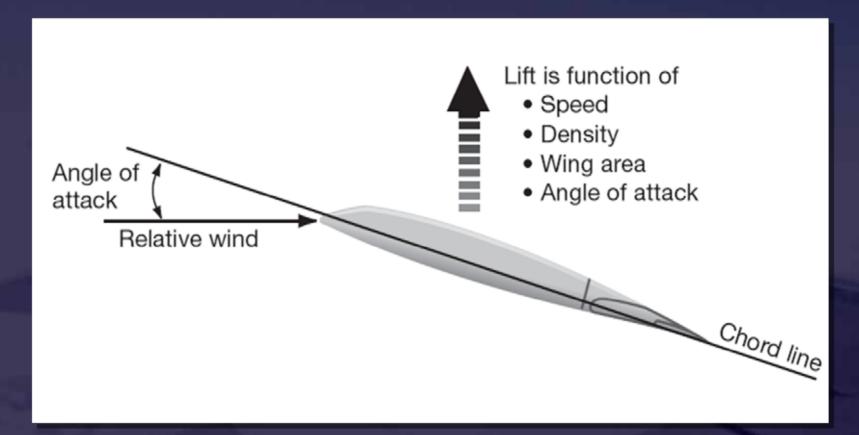
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Aerodynamic Review



Lift = $\frac{1}{2} \rho.S.V^2.CI$

ρ: air density

S: wing surface

V: CAS

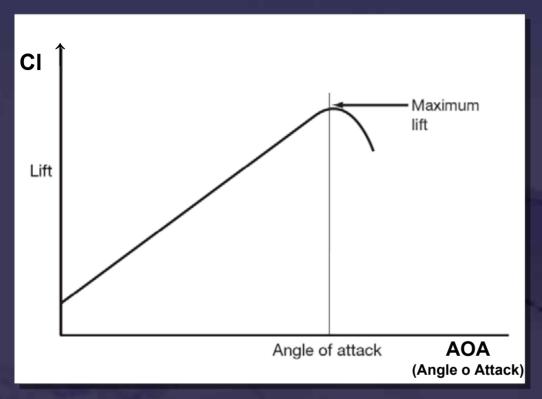
CI: lift coefficient







Lift Coefficient

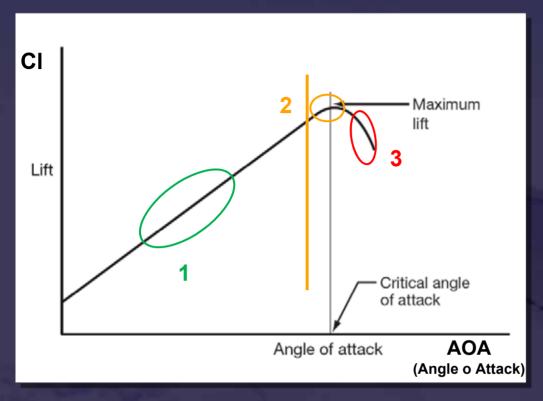


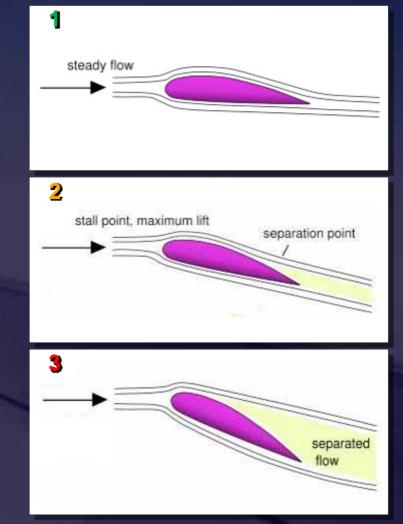
CI has a direct relationship to AoA

For a given configuration, a given speed and a given altitude, The Lift is only linked to AoA



Stall





- CI has a direct relationship to AoA
- At a given AoA, the air flow separates from the profile, the CI drops P The wing profile is stalled



For a given aircraft configuration and speed, An aircraft stalls for a given AoA









Stall

Stall is an AoA problem only

For a given load factor, the loss of speed leads to increase the AoA, which could reach the stall AoA

> Stall is an AoA problem only It is NOT directly a speed issue







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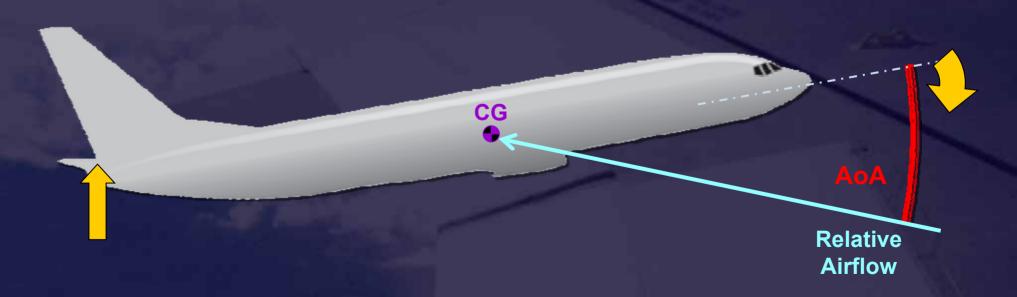






The pitch control is a direct AoA command

- ▶ The elevators control DIRECTLY the AoA
- A nose down command has an IMMEDIATE effect: an AoA decrease



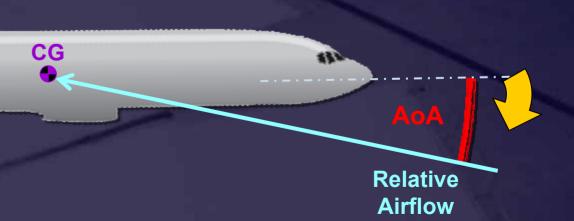






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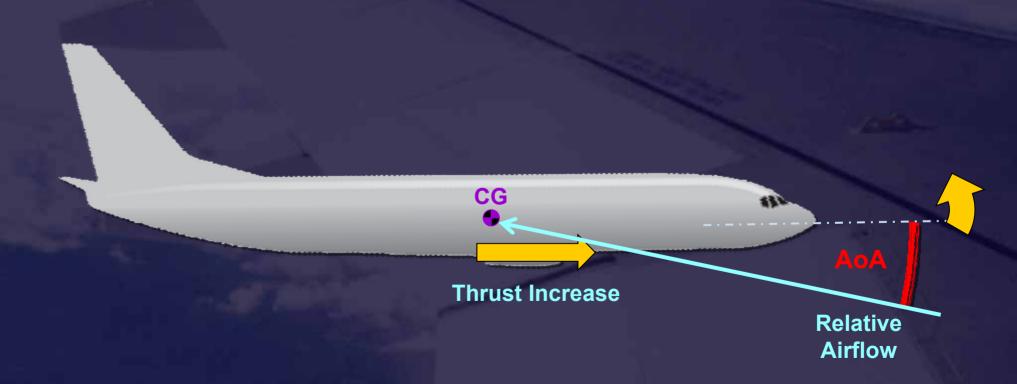






Aircraft with engine below aircraft CG

PThrust has a significant pitch effect



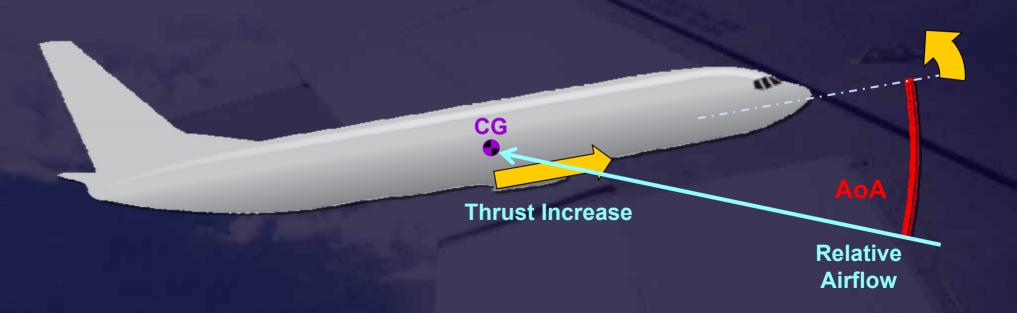




Aircraft with engine below aircraft CG

PThrust has a significant pitch effect

Thrust increase ⇒ AoA increase

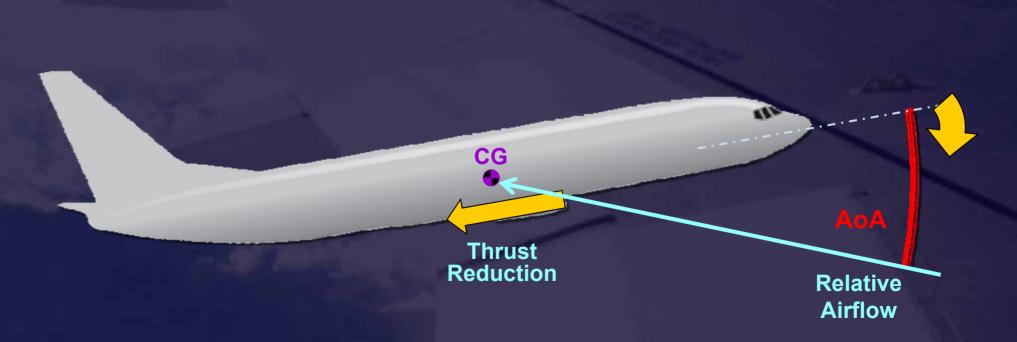






Aircraft with engine below aircraft CG

PThrust has a significant pitch effect





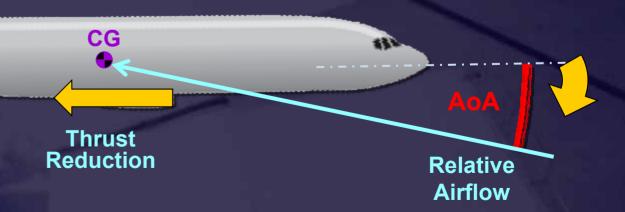




Aircraft with engine below aircraft CG

PThrust has a significant pitch effect

Thrust reduction ⇒ AoA decrease



14/23







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Stall vs. Approach To Stall

- The traditional APPROACH TO STALL training is characterized by a controlled deceleration to stall warning, followed by a power recovery with minimum altitude loss
- The difference between an APPROACHING STALL and an ACTUAL STALL is not easy to determine, even for specialists
- In numerous accidents, the APPROACH TO STALL procedure was applied whereas the aircraft was actually stalled







Approach To Stall Recovery

- Classical Approach to stall procedure focuses:
 - On thrust application
 - Minimum loss of altitude
- It is NOT appropriate for EVERY stall condition:
 - Possible inability to reduce AoA with the TOGA thrust application
 - Recovery may require thrust reduction
 - Recovery from a stall may require altitude loss

A single procedure has to be defined focusing on AoA reduction





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Stall Recovery

As soon as any stall indication is recognized

- "NOSE DOWN PITCH CONTROL.....APPLY":
 - AoA must be reduced to regain lift:
 - Apply nose down pitch order on the side-stick
 - If needed, reduce thrust in case of lack of pitch down authority
 - Ensure wings are level

AoA comes first...







Stall Recovery

When out of stall

- INCREASE ENERGY
 - Thrust increase smoothly as needed

. Immediate maximum thrust application upon stall recognition is not appropriate

. Thrust has an adverse effect on AoA for aircraft with under wing mounted engines

. Due to the engine spool up time, the aircraft speed increase that results from thrust increase,

is slow and does not enable to reduce the AOA instantaneously

- Ensure that speed brakes are retracted
- Recover smoothly the initial flight path

AoA comes first, energy second







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New Procedure

Spirit of what is the new procedure

- One single procedure to cover ALL stall conditions
- Get rid of TOGA as first action
- Focus on AoA reduction





New Procedure

 The stall recovery procedure is available on the Airbus World portal and will be dispatched to operators as follows:

A318/319/320/321 aircraft:

- FCOM volume 3 Temporary Revision number 323-1
- QRH Temporary Revision number 727-1

► A330 aircraft:

- FCOM volume 3 Temporary Revision number 552-1
- QRH Temporary Revision number 353-1

▶ A340 aircraft:

- FCOM volume 3 Temporary Revision number 512-1 (A340-200/-300)
- FCOM volume 3 Temporary Revision number 513-1 (A340-500/-600)
- QRH Temporary Revision number 369-1

► A380 aircraft:

- FCOM Procedures / Non-ECAM Abnormal and Emergency Procedures / **Operating Techniques**
- In addition, the Flight Crew Training Manuals will be revised accordingly









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Stall

